ATSRAC

Lessons Learned

Miami, January 17 - 18, 2001

Introduction

share "lessons learned" by the OEM's and the fleet, which address aspects of design and installation according to the following plan: The FAA & ATSRAC tasked the OEM's to Identify, document and

- Situation from each of the significant items from Non-Intrusive aircraft survey.
- situation as how to address these conditions. Detailed focus of the noted conditions from aircraft survey, and



Situation from each of the significant items from Non-Intrusive aircraft survey

of A300 aeroplanes, five individual items were deemed significant any of the inspected aeroplanes From the survey of aeroplane wiring installed on representative examples None were safety of flight concerns that would require immediate action on

additional engineering analysis was conducted to propose solutions as processes and/or Service Bulletins necessary via either maintenance enhanced inspection guidelines & However, for reasons of repeat occurrences in the same general area,



Situation from each of the significant items from Non-Intrusive aircraft survey (Cont'd).

- Inspection SB is on creation process with compliance Recommended - ESPM 20 Repair section will be revised to provide enhanced technical process for improved gripping of the clamping attachment when found loose - EspM 20 Repair section will be revised to provide enhanced technical process for improved enhanced inspection guidelines as per Task 3 report will be introduced in MPD section introduction EspM 20 Repair section will be revised to provide enhanced technical process for improved gripping of the clamping attachment when found loose - EspM 20 Repair section will be revised to provide enhanced technical process for improved gripping of the clamping attachment when found loose - EspM 20 Repair section will be revised to provide enhanced technical process for improved gripping of the clamping attachment when found loose - Standard Manual (SM) will be revised for introduction of std parts interchangeability Upon Design Modification approval (on process), a Mod SB will be launched with compliance enhanced triveted brackets Illustrated Parts Catalog (IPC) will reflect the latest installation configuration as per MOD/SB enhanced riveted brackets.	ellipodillett & tepotitis process to the clevision.
ISB: ESP ecd MPE ESP ESP ecd MSE	b. Bracket - Opon Design Modification approval (on process), a mod SB will be la unstuck at panel Recommended, to introduce riveted brackets. - Illustrated Parts Catalog (IPC) will reflect the latest installation configu
ISB : ESP ecd MPE ESP ESP SM ecd ecd ecd	
ISB ESP ESP ESP SM ecd ecd	embodiment & reporting process for IPC revision.
ISB: ESP ESP ESP SM AMPE	panel 811VU Recommended, to introduce additional bundle protection and attachment.
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	- Inspection SB is on creation process with compliance Recommended
	Action plan on process



survey, and situation as how to address these conditions Detailed focus of the noted conditions from aircraft

conditions have been found as the most predominant: From the aircraft surveys (both Non-Intrusive and Intrusive), the following

- Fluid or chemical contamination
- Inadequate wire repair (unapproved parts)
- Significant dust/lint build-up & Debris accumulation
- Wire/bundle excessive tension
- Wire Clamp condition, sizing, spacing /bundle slacking or sagging



Detailed focus of the noted conditions from aircraft conditions (Cont'd) survey, and situation as how to address these

Most predominant conditions (cont'd):

- Bundle segregation
- Inadequate clearance to structure
- Incorrect hardware build-up (at terminal block)
- Condition and security of ground point
- Damaged backshell

appendix to this presentation. pertaining to each of the above conditions are provided in a separate Evaluation and recommendation, as well as (Airbus) document references



Conclusion

From ATSRAC initiatives, lessons have been drawn from aircraft inspections.

Actions to develop improvements have been launched.

Progressive implementation is scheduled in the course of 2001.

All the aircraft operators will be informed of these results through an Operator Information Telex (ECD 1st quarter 2001).



- APPENDIX -

Detailed focus of the noted conditions from aircraft survey

Conditions

Evaluation

and

Recommendations

Document Ref

Condition and security of ground point	Incorrect hardware build- up (at terminal block stud)	Inadequate clearance to structure
Similar to the discussion concerning incorrect hardware build-up. Additional consideration is to be provided to the coating or varnish protection to be applied or re-applied on the ground connection hardware to avoid corrosion.	Deterioration/deformation of terminal lug contact surfaces can result in electrical contact resistance, and overheat of the connections. Deformations at the lug-to-crimp barrel area can initiate cracks, and possibly lug rupture with vibrations. This can possibly result into intermittent arcing. The use of incorrect screw, washer, nuts can create corrosion due to dissymetric materials as well as incorrect selection of terminal lug. Wrong orientation of the terminal lugs can create condition for inadequate clearance to adjacent wires and/or environment. If the connection is over-tightened or somewhat loose (i.e. not in compliance with specified tightening/torque values), this can create an additional condition to initiate degradation.	Originated from incorrect orientation of clamps, missing or broken or unsecured clamps/ties, bundle slacking, sagging, wire bundle displacement, combined with maintenance traffic, vibrations, sometimes modification (e.g. cabin layout, conversion),
Electrical bonding types, description, operation, protection and testing are documented in ESPM	Much information pertaining to terminal lug selection, assembly, stacking processes is available in ESPM. Torque values are provided in both ESPM and AMM.	Lay-out of bundle, routing along structure, piping, bundle slack or sag limit criteria, bend radius, attachment, drip loops, routes categories, segregation rules are available in ESPM.
ESPM 20-28-20 to 20-28-50	ESPM 20-48-11 20-48-12 20-48-13 20-48-18 20-44-52 AMM 20-21-xx	ESPM 20-60-00

